

FAA COPY

DEPARTMENT OF TRANSPORTATION  
FEDERAL AVIATION ADMINISTRATION

Form Approved  
Budget Bureau No. 04-R060.1

**MAJOR REPAIR AND ALTERATION**  
(Airframe, Powerplant, Propeller, or Appliance)

FOR FAA USE ONLY

OFFICE IDENTIFICATION

AGL 0100 #8

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE <b>Beech</b>	MODEL <b>95-1355</b>
	SERIAL NO. <b>TC-1730</b>	NATIONALITY AND REGISTRATION MARK <b>N7384R</b>
2. OWNER	NAME (As shown on registration certificate) <b>Boyd E. Hart Co.</b>	ADDRESS (As shown on registration certificate) <b>34670 Lakeland Blvd. Eastlake, Ohio 44094</b>

3. FOR FAA USE ONLY

THIS DATA IDENTIFIED HEREIN COMPLIES WITH APPLICABLE AIRWORTHINESS REQUIREMENTS AND IS APPROVED ONLY FOR THE ABOVE DESCRIBED AIRCRAFT. SUBJECT TO CONFORMITY INSPECTION AS AUTHORIZED IN FAR 43.9. APPROVING INSPECTOR *[Signature]* DATE **29 MAY 79**

4. UNIT IDENTIFICATION				5. TYPE	
UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	As described in item 1 above				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS <b>Ohio Aviation Company 355 Richmond Road Cleveland, Ohio 44143</b>	B. KIND OF AGENCY	C. CERTIFICATE NO. <b>1195</b>
	U.S. CERTIFICATED MECHANIC	
	FOREIGN CERTIFICATED MECHANIC	
	<input checked="" type="checkbox"/> CERTIFICATED REPAIR STATION	
	MANUFACTURER	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE <b>JUNE 4th 1979</b>	SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Ronald P. Perrett</i> 2247470
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7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  APPROVED  REJECTED

BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE	<input checked="" type="checkbox"/> REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	
DATE OF APPROVAL OR REJECTION <b>6-4-79</b>	CERTIFICATE OR DESIGNATION NO. <b>1195</b>	SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Ronald P. Perrett</i> 2247470 <b>MAY 23 1979</b>		

## NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

### RNAV INSTALLATION AND IFR CERTIFICATION:

NEW INSTALLATION OF FOSTER AIR DATA RNAV SYSTEM AD-511 WITH ALPHA STEERING ADAPTER 51ASA.

1. Reference Foster AD-511 sn 1195, with 51ASA sn 594 RNAV System as shown on aircraft paperwork dated 6-4-79, interconnected with KING KN-65 DME and number one (as the primary) KX-170B KING NAV receiver, King KI214 NAV converter, and number two (as a secondary) KING KX-170B Navigation receiver and King KI201-C converter. The RNAV system has been bench checked and ramp checked in accordance with AC-90-45A, appendix "A" Par2,3b, and 3c, and has been found acceptable per AC-90-45A appendix "D" table 2,3, and 4, on 6-4-79?
2. The wiring and interconnect has been accomplished per FOSTER AIR DATA AD-511 ALPHA SYSTEM installation manual. The installation has further been accomplished and checked in accordance with FAA AC-43-13-1A and 43-13-2A. Necessary additions & updating to equipment list & weight & balance data have been accomplished, with log book entries to reflect same as of 6-4-79.
3. Aflight check of the RNAV system installed in this aircraft as referenced in this FAA 377 Form have been conducted in the Cleveland Ohio area as of 6-4-79.
4. Multiple direction waypoint crossings over a known landmark have been accomplished to insure proper operation, and that track errors and approach accuracies are within acceptable IFR parameters, as of 6-4-79 Further, two (2) RNAV approaches were made per published approach data as of 6-4-79. First to Cuyahoga County CGF RNAV-23 approach, second to Cleveland Hopkins CLE RNAV-36LEFT approach.
5. As of 6-4-79 all RNAV approach and track errors were found well within the accuracies established in tables 2,3, and 4 of appendix "A", Advisory Circular 90-45A.
6. If any component of the RNAV system is repaired or recalibrated, the RNAV system's accuracy should be rechecked to show compliance with original certification. All maintenance of the RNAV system will be performed as outlined in FAA AC-90-45A appendix "E".
7. Removal of the AD-511 computer does NOT render either the #1 or #2 existing NAVIGATION system's inoperative.
8. Applicable placards have been installed as of 6-4-79.
9. This aircraft as log book entry dated 6-4-79 reflects, is hereby approved for ENROUTE-TERMINAL & APPROACH modes of IFR OPERATION.
10. END.

**MAJOR REPAIR AND ALTERATION**  
(Airframe, Powerplant, Propeller, or Appliance).

FOR FAA USE ONLY  
OFFICE IDENTIFICATION

AGI 0400 #6

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE Beechcraft	MODEL B-55
	SERIAL NO. TC1730	NATIONALITY AND REGISTRATION MARK N7384R
2. OWNER	NAME (As shown on registration certificate) Boyd E. Hart Company	ADDRESS (As shown on registration certificate) Eastlake, Ohio 34670 Lakeland Blvd. 44094

3. FOR FAA USE ONLY

4. UNIT IDENTIFICATION

UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
				REPAIR	ALTERATION
AIRFRAME	***** (As described in item 1 above) *****				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS Ohio Aviation Company 355 Richmond Road Cleveland, Ohio 44143	B. KIND OF AGENCY	C. CERTIFICATE NO. CRS1195
	<input type="checkbox"/> U.S. CERTIFICATED MECHANIC	
	<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC	
	<input checked="" type="checkbox"/> CERTIFICATED REPAIR STATION	
	<input type="checkbox"/> MANUFACTURER	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above, and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE January 11, 1980	SIGNATURE OF AUTHORIZED INDIVIDUAL <i>William C. Astalos</i> William C. Astalos
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7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  APPROVED  REJECTED

BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE	<input checked="" type="checkbox"/> REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	
DATE OF APPROVAL OR REJECTION Jan. 11, 1980	CERTIFICATE OR DESIGNATION NO. CRS1195	SIGNATURE OF AUTHORIZED INDIVIDUAL <i>William C. Astalos</i> William C. Astalos		

JAN 21 1980

# NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

1. Installed RCA weatherscout II Radar system. Installation was performed in accordance with manufacturers recommendations; using FAA AC-43-13-1A CH 11 and 43-13-2A CH 1 and 2 as a guide.
2. Installed Norton Radome Kit number 4083X (STC SA64GL) in accordance with manufacturers recommendations and prints, again using FAA AC-43-13-1A and 43-13-2A.
3. Weight and balance has been changed to reflect this installation.
4. A flight test was conducted to assure this installation does not compromise airworthiness.

-----End-----

**MAJOR REPAIR AND ALTERATION**  
**(Airframe, Powerplant, Propeller, or Appliance)**

FOR FAA USE ONLY  
OFFICE IDENTIFICATION

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE Beech	MODEL 95-B55
	SERIAL NO. TC-1730	NATIONALITY AND REGISTRATION MARK N7384R
2. OWNER	NAME (As shown on registration certificate) Hempt Max C	ADDRESS (As shown on registration certificate) 245 Hempt Road Mechanicsburg PA 17055

3. FOR FAA USE ONLY

4. UNIT IDENTIFICATION

5. TYPE

UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
				REPAIR	ALTERATION
AIRFRAME	♦♦♦♦♦♦♦♦♦♦♦♦♦♦♦♦♦♦♦♦ (As described in item 1 above) ♦♦♦♦♦♦♦♦♦♦♦♦♦♦♦♦♦♦♦♦				XX
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
B. F. GOODRICH - AIRCRAFT SERVICE CENTER AKRON-CANTON REGIONAL AIRPORT NORTH CANTON, OHIO 44720 F.A.A. CERTIFIED REPAIR STATION #1524	<input type="checkbox"/> U.S. CERTIFICATED MECHANIC	#1524 Limited Airframe
	<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC	
	<input checked="" type="checkbox"/> CERTIFICATED REPAIR STATION	
	<input type="checkbox"/> MANUFACTURER	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE 3-7-84	SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Norman P. Ashburn</i> Norman P Ashburn
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7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  APPROVED  REJECTED

BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE	<input checked="" type="checkbox"/> REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	
DATE OF APPROVAL OR REJECTION 3-7-84	CERTIFICATE OR DESIGNATION NO. #1524	SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Norman P. Ashburn</i> Norman P Ashburn		

3-1-81

#T2S#

# NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installed Airborne pneumatic surface deicer system kit no 708A s/n 0841 in accordance with STC SA799CE and installation manual.  
Equipment list revised  
Weight and Balance Computed.

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1. A.1.1. CERTIFIED REPAIR STATION #T2S#	1. A.1.1.2. ADDRESS	1. A.1.1.3. CITY	1. A.1.1.4. STATE	1. A.1.1.5. ZIP
ARCON-CANTON REGIONAL AIRPORT	ARCON-CANTON REGIONAL AIRPORT	ARCON-CANTON REGIONAL AIRPORT	OHIO	44350
1. A.1.1.6. AGENCY NAME AND ADDRESS	1. A.1.1.7. KIND OF AGENCY	1. A.1.1.8. CERTIFICATE NO.		

A.1.1.1. AIRLINE	A.1.1.2. DATE	A.1.1.3. MODEL	A.1.1.4. SERIAL NO.	A.1.1.5. IDENTIFICATION	
				REGISTRATION MARK	TYPE
					XX

1. A.1.1.9. NAME (Last, first, middle initial)	1. A.1.1.10. ADDRESS (Street or telephone address)
Henry Max C	2502 North Road Mesa, Arizona 85202
1. A.1.1.11. SERIAL NO.	1. A.1.1.12. NATIONALITY AND REGISTRATION MARK
LC-1130	N1381B
1. A.1.1.13. MAKE	1. A.1.1.14. MODEL
Boeing	32-322

1. A.1.1.15. (Signature, Position, Title or Address) WALTON BERRY AND ASSOCIATES FEDERAL AVIATION ADMINISTRATION DEPARTMENT OF TRANSPORTATION	1. A.1.1.16. OFFICE IDENTIFICATION
	1. A.1.1.17. DATE
	1. A.1.1.18. SIGNATURE

ADDITIONAL SHEETS ARE ATTACHED

**MAJOR REPAIR AND ALTERATION**  
**(Airframe, Powerplant, Propeller, or Appliance)**

FOR FAA USE ONLY  
OFFICE IDENTIFICATION

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE <i>BEECH</i>	MODEL <i>95-B55</i>
	SERIAL NO. <i>TC1730</i>	NATIONALITY AND REGISTRATION MARK <i>N 42 HF</i>
2. OWNER	NAME (As shown on registration certificate) <i>Hunt Farms</i>	ADDRESS (As shown on registration certificate) <i>205 Birch Road Camp Hill, PA 17011</i>

3. FOR FAA USE ONLY

The data identified herein complies with the applicable airworthiness requirements of appropriate subtitles of TSO C-20 and is approved for duplication on identical aircraft heaters of the B-1500 thru the B-4050 series, when alteration is accomplished by the original modifier.

AGL-GADO-18 Principal Maintenance Inspector

*Robert E. Kendig*

4. UNIT IDENTIFICATION				5. TYPE	
UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	♦♦♦♦♦♦♦♦♦♦♦♦♦♦♦♦ (As described in item 1 above) ♦♦♦♦♦♦♦♦♦♦♦♦♦♦♦♦				
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				X

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS  C & D Airmotive 302 Post Road Buchanan, Michigan 49107	B. KIND OF AGENCY		C. CERTIFICATE NO.  C-18-24
	U.S. CERTIFICATED MECHANIC		
	FOREIGN CERTIFICATED MECHANIC		
	X CERTIFICATED REPAIR STATION		
		MANUFACTURER	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE  05-23-86	SIGNATURE OF AUTHORIZED INDIVIDUAL  <i>Sharon K. Thomas</i>
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7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  APPROVED  REJECTED

BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE X	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	
DATE OF APPROVAL OR REJECTION 05-23-86	CERTIFICATE OR DESIGNATION NO. C18-24	SIGNATURE OF AUTHORIZED INDIVIDUAL  <i>Sharon K. Thomas</i>		

## NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

### 8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

1. The combustion heater tube assembly installed in the heater identified in Block #4, has been approved for installation based on data and tests listed below in accordance with applicable subtitles of the Society of Automotive Engineers, Inc., Aeronautical Standards AS-143B, which is the basic standards for TSO C-20.
2. In accordance with Subtitle 4.5 of AS-143B, this combustion tube assembly will be marked with the C & D Airmotive logo near the combustion air pressure switch attachment point, via electro chemical process. This marking can readily be seen with the heater shroud installed, and by visual inspection after installation in the aircraft.
3. The unit is constructed of stainless steel type 309.029, which is equivalent to AMS 5540, required by subtitle 5.4.1 of AS-143B.
4. Equivalent unit was operated in excess of 500 hours per subtitle 5.4.3 and 5.4.4 of AS-143B, with no deterioration as determined by subsequent analysis.
5. Tests were conducted which exceeded subtitle 6.1.6 and 6.1.6.1 of AS-143B with no deterioration incurred.
6. Upon completion of the previously listed subtitles, a pressure decay test required under paragraph "e" of AD 82-07-03 was conducted, and the results exceeded these requirements.

NOTE: The installation of the combustion tube does not relieve any responsibility for compliance with AD 82-07-03.

END

NOTE :

INSTALLED IN N42HF, BEECH

15 Oct 86. *A. C. C. C.*

Hobbs Inc 115 *17400917IA*

ADDITIONAL SHEETS ARE ATTACHED



U.S. Department  
of Transportation  
**Federal Aviation  
Administration**

**MAJOR REPAIR AND ALTERATION  
(Airframe, Powerplant, Propeller, or Appliance)**

Form Approved  
OMB No. 2120-0020

**For FAA Use Only**

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make Beechcraft	Model 95-B55
	Serial No. TC-1730	Nationality and Registration Mark N-42HF
2. Owner	Name (As shown on registration certificate) Bruhn, Thomas W.	Address (As shown on registration certificate) 351 Ushers Road Ballston Lake, NY 12012

**3. For FAA Use Only**

**4. Unit Identification**

**5. Type**

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

**6. Conformity Statement**

A. Agency's Name and Address MURRAY AVIONICS, INC. 19 AIRPORT ROAD SCOTIA, NY 12302	B. Kind of Agency <input type="checkbox"/> U.S. Certificated Mechanic <input checked="" type="checkbox"/> Foreign Certificated Mechanic <input checked="" type="checkbox"/> Certificated Repair Station <input type="checkbox"/> Manufacturer	C. Certificate No. # AL1R012K
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D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 6/16/92	Signature of Authorized Individual <i>Richard B. Murray</i>
-----------------	--

**7. Approval for Return To Service**

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  APPROVED  REJECTED

BY	FAA Flt. Standards Inspector	Manufacturer	Inspection Authorization	Other (Specify)
	FAA Designee	<input checked="" type="checkbox"/> Repair Station	Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection 6/16/92	Certificate or Designation No. #AL1R012K	Signature of Authorized Individual <i>Richard B. Murray</i>
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### NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

#### 8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installed WX-900 Stormscope system.

All equipment installed in accordance with instruction provided by the equipment manufacturers instruction manual (WX-900 manual dated JUL 1991 and adapted to the aircraft IAW AC 43.13-2A fig. 2.1 and fig. 3.6.

Aircraft equipment list and weight and balance revised to reflect this change.

Continuous electrical running load computed and found to be within 80% of the charging system capacity.

Static load test performed and exceeds load factors as recommended in AC 43.13-2A para. 1.2d.

Unit operates satisfactorily and does not interfere with any other systems. Unit is not connected to any other system.

\*\*\*\*\*END\*\*\*\*\*

A. Agency's Name and Address		MURRAY AVIONICS, INC. 19 AIRPORT ROAD SCOTIA, NY 13302	
B. Kind of Agency	C. Certificate No.	D. Manufacturer	E. Part Number
U.S. Certified Mechanic			
Foreign Certified Mechanic			
Canadian Approved Repair Station			
F. Signature of Authorized Individual			
<i>[Signature]</i>			
G. Date of Approval or Refection			
11/19/92			
H. Signature of Authorized Individual			
<i>[Signature]</i>			
I. Date of Approval or Refection			
11/19/92			
J. Signature of Authorized Individual			
<i>[Signature]</i>			
K. Date of Approval or Refection			
11/19/92			
L. Signature of Authorized Individual			
<i>[Signature]</i>			
M. Date of Approval or Refection			
11/19/92			
N. Signature of Authorized Individual			
<i>[Signature]</i>			
O. Date of Approval or Refection			
11/19/92			
P. Signature of Authorized Individual			
<i>[Signature]</i>			
Q. Date of Approval or Refection			
11/19/92			
R. Signature of Authorized Individual			
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S. Date of Approval or Refection			
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11/19/92			
V. Signature of Authorized Individual			
<i>[Signature]</i>			
W. Date of Approval or Refection			
11/19/92			
X. Signature of Authorized Individual			
<i>[Signature]</i>			
Y. Date of Approval or Refection			
11/19/92			
Z. Signature of Authorized Individual			
<i>[Signature]</i>			
AA. Date of Approval or Refection			
11/19/92			
AB. Signature of Authorized Individual			
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AC. Date of Approval or Refection			
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AD. Signature of Authorized Individual			
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AE. Date of Approval or Refection			
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BK. Date of Approval or Refection			
11/19/92			
BL. Signature of Authorized Individual			
<i>[Signature]</i>			
BM. Date of Approval or Refection			
11/19/92			
BN. Signature of Authorized Individual			
<i>[Signature]</i>			
BO. Date of Approval or Refection			
11/19/92			
BP. Signature of Authorized Individual			
<i>[Signature]</i>			
BQ. Date of Approval or Refection			
11/19/92			
BR. Signature of Authorized Individual			
<i>[Signature]</i>			
BS. Date of Approval or Refection			
11/19/92			
BT. Signature of Authorized Individual			
<i>[Signature]</i>			
BU. Date of Approval or Refection			
11/19/92			
BV. Signature of Authorized Individual			
<i>[Signature]</i>			
BW. Date of Approval or Refection			
11/19/92			
BX. Signature of Authorized Individual			
<i>[Signature]</i>			
BY. Date of Approval or Refection			
11/19/92			
BZ. Signature of Authorized Individual			
<i>[Signature]</i>			
C0. Date of Approval or Refection			
11/19/92			

Additional Sheets Are Attached



US Department of Transportation  
Federal Aviation Administration

## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved  
OMB No. 2120-0020

**For FAA Use Only**  
Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

<b>1. Aircraft</b>	Make Beechcraft	Model 95-B55
	Serial No. TC-1730	Nationality and Registration Mark N-42HF
<b>2. Owner</b>	Name (As shown on registration certificate) Bruhn, Thomas W.	Address (As shown on registration certificate) 351 Ushers Road Ballston Lake, NY 12012

**3. For FAA Use Only**

The ~~alteration~~ / data identified herein complies with the applicable airworthiness requirements and is approved for the above described aircraft, subject to conformity inspection by a person authorized in FAR 43, Section 43.7

**AEA - FSDO - 1**

11-23-92   
Date                      FAA Inspector

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

**6. Conformity Statement**

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
MURRAY AVIONICS, INC. 19 AIRPORT ROAD SCOTIA, NY                      12302	<input type="checkbox"/> U.S. Certificated Mechanic	#ALLR012K
	<input type="checkbox"/> Foreign Certificated Mechanic	
	<input checked="" type="checkbox"/> Certificated Repair Station	
	<input type="checkbox"/> Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 8/21/92	Signature of Authorized Individual 
-----------------	--

**7. Approval for Return To Service**

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  APPROVED     REJECTED

BY	FAA Fit. Standards Inspector	Manufacturer	Inspection Authorization	Other (Specify)
	FAA Designee	<input checked="" type="checkbox"/> Repair Station	Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection 11/24/92	Certificate or Designation No. ALLR012K	Signature of Authorized Individual 
---	--	--

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

**8. Description of Work Accomplished**

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)  
Removed Foster 511 R-Nav.

Installed Northstar M-600 G.P.S. System.

All equipment installed in accordance with instruction provided by the aircraft radio manufacturers instruction manual (GM-450--Rev. A) and adapted to the aircraft IAW AC 43.13-2A fig. 2.1 and fig. 3.6, paragraphs 23 and 27.

Aircraft equipment list and weight and balance revised to reflect this change.

Continuous electrical running load computed and found to be within 80% of the 50 amp alternators. Unit is powered from a 5 amp breaker in the breaker panel placarded G.P.S.

Static load test performed and exceeds load factors as recommended in AC 43.13-2A para. 1.2d.

Unit operates satisfactorily and does not interfere with any other systems. Unit is integrated with the HSI through an interface switch, with a G.P.S./VOR annunciator mounted on the instrument panel. Unit is placarded : "G.P.S. NOT APPROVED FOR IFR".

\*\*\*\*\*END\*\*\*\*\*

Additional Sheets Are Attached



US Department of Transportation  
Federal Aviation Administration

## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved  
OMB No. 2120-0020

**For FAA Use Only**  
Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

<b>1. Aircraft</b>	Make Beechcraft	Model 95-B55
	Serial No. TC-1730	Nationality and Registration Mark N42HF
<b>2. Owner</b>	Name (As shown on registration certificate) Bruhn, Thomas W.	Address (As shown on registration certificate) 351 Ushers Road Ballston Lake N. Y. 12012

### 3. For FAA Use Only

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

### 6. Conformity Statement

<b>A. Agency's Name and Address</b> United Beechcraft Republic Airport Hangar B East Farmingdale N. Y. 11735	<b>B. Kind of Agency</b> <input type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input checked="" type="checkbox"/> Certificated Repair Station <input type="checkbox"/> Manufacturer	<b>C. Certificate No.</b> CRS FV1R154K AIRFRAME CLASS 3
--	---	--

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 7-30-93	Signature of Authorized Individual 
-----------------	--

### 7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  APPROVED  REJECTED

BY	FAA Fit. Standards Inspector		Manufacturer	Inspection Authorization	Other (Specify)
	FAA Designee	<input checked="" type="checkbox"/>	Repair Station	Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection 7-30-93	Certificate or Designation No. CRS FV1R154K	Signature of Authorized Individual 
--	--	--

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

**8. Description of Work Accomplished**

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

These Propellers Model PHC-C2YF-2CHF Left S/N 6374, Right S/N 812 have been Balanced in Accordance with FAA Approved Chadwick Helmuth AW-9511-2 Manual. Weight and Balance is negligable and a log book entry has been made. Reference United Beechcraft Inc. Work Order No. 15558

END

2. Type		4. Unit Identification		5. For FAA Use Only	
Alteration	Repair	Serial No.	Model	Make	Unit
X					PROPPELLER
					POWERPLANT
					APPLIANCE

A. Agency Name and Address United Beechcraft Republic Airport Hanger B East Farmingdale N. Y. 11735		B. Kind of Agency U.S. Certified Mechanic Foreign Certified Airframe X Certified Repair Station Manufacturer		C. Certificate No. CRS FVLR15AK AIRFRAME CLASS 3
B. I certify that the repair and/or alteration described in item 4 above and described on the reverse of attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.				
Signature of Authorized Individual 		Date 7-30-93		
T. Approval for Return to Service In accordance with the following given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED				
Signature of Authorized Individual		Inspection Substation	Manufacturer	FAA Part Standards Inspector
Signature of Authorized Individual		Person Approved by Transport Canada Airworthiness Group	Repair Station	FAA Designee
Signature of Authorized Individual		Certificate or Designation No.		Date of Approval or Expiration
		CRS FVLR15AK		7-30-93

Additional Sheets Are Attached



US Department  
of Transportation  
Federal Aviation  
Administration

## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved  
OMB No. 2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B. and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

<b>1. Aircraft</b>	Make <b>Beechcraft</b>	Model <b>Beech Baron B55</b>
	Serial No. <b>TC-1730</b>	Nationality and Registration Mark <b>N42HF</b>
<b>2. Owner</b>	Name (As shown on registration certificate) <b>Humes Thomas M.</b>	Address (As shown on registration certificate) <b>1785 Doolittle Ct Daytona Beach, FL. 321246757</b>

### 3. For FAA Use Only

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				<b>XXX</b>
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

### 6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
<b>DBM 3127 Creve Couer Rd. Hangar E-7 Creve Couer, MO. 63146</b>	<input type="checkbox"/> U.S. Certified Mechanic	<b>SH2R161L</b>
	<input type="checkbox"/> Foreign Certified Mechanic	
	<input checked="" type="checkbox"/> Certified Repair Station	
	<input type="checkbox"/> Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date <b>02/23/2000</b>	Signature of Authorized Individual  <b>James Klug</b>
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### 7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  APPROVED  REJECTED

	FAA Flt. Standards Inspector	Manufacturer	Inspection Authorization	Other (Specify)
	FAA Designee <input checked="" type="checkbox"/>	Repair Station <input checked="" type="checkbox"/>	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection <b>02/23/2000</b>	Certificate or Designation No. <b>SH2R161L</b>		Signature of Authorized Individual  <b>James Klug</b>	

# NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

## 8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Performed the following modifications and or replacements as per the STC numbers listed below.

WINDSHIELD AS PER STC SA993NE

See log book entry and new empty weight for this date.

Additional Sheets Are Attached

US Department of Transportation Federal Aviation Administration	<b>MAJOR REPAIR AND ALTERATION</b> <b>(Airframe, powerplant, Propeller, or Appliance)</b>	Form Approved OMB No.2120-0020 For FAA Use Only Office Identification
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INSTRUCTIONS: Print or type all entries. See FAR43.9, FAR 43 Appendix B. and AC43.9-1 (or subsequent revision thereof) for instructions and disposition of this form This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make <b>BEECH</b>	Model <b>95-B55</b>
	Serial No. <b>TC-1730</b>	Nationality and Registration Mark <b>N42HF</b>
2. Owner	Name (As shown on registration certificate) <b>HUMES THOMAS M</b>	Address (As Shown on registration certificate) <b>1785 DOOLITTLE CT DAYTONA BEACH FL 32124-6757</b>

**3. For FAA Used Only**

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	-----	(As described in item 1 above)	-----		<b>XXX</b>
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement		
A. Agency's Name and Address <b>Ernest W. Colbert COLEMILL ENTERPRISES INC. Post Office Box 60627 Nashville, TN 37206</b>	B. Kind of Agency <input checked="" type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input type="checkbox"/> Certificated Repair Station <input type="checkbox"/> Manufacturer	C. Certificate No. <b>AP445788</b>

D. I certify that the repair and/or alteration made to units(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date <b>2-20-01</b>	Signature of Authorized Individual  <b>Ernest W. Colbert</b>
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**7. Approval for Return To Service**

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  APPROVED  REJECTED

By	FAA Fit. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/> Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station	Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection <b>2-20-01</b>	Certificate or Designation No. <b>AP2178607IA</b>	Signature of Authorized Individual  <b>KARL M. CRIST</b>
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US Department of Transportation  
Federal Aviation Administration

**MAJOR REPAIR AND ALTERATION**  
**(Airframe, Powerplant, Propeller, or Appliance)**

Form Approved  
OMB No. 2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make Beechcraft	Model 95-B55
	Serial No. TC-1730	Nationality and Registration Mark N42HF
2. Owner	Name (As shown on registration certificate) Humes, Thomas M.	Address (As shown on registration certificate) 1785 Doolittle Ct. Daytona Beach, FL 32124-6757

3. For FAA Use Only

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	_____ (As described in Item 1 above) _____				XXX
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
Thomas M. Humes 1785 Doolittle Ct. Daytona Beach, FL 32124	<input checked="" type="checkbox"/> U.S. Certified Mechanic	2635991
	<input type="checkbox"/> Foreign Certified Mechanic	
	<input type="checkbox"/> Certified Repair Station	
	<input type="checkbox"/> Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 16 Apr 2001	Signature of Authorized Individual 
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7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  APPROVED  REJECTED

BY	FAA Flt. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/> Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 4/16/01		Certificate or Designation No. 1191065	Signature of Authorized Individual 	

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

1. Removed Janitrol heater P/N 81D94-1 and pressure differential switch P/N A42D36. Installed rebuilt Janitrol heater Model B4500, P/N FR81D94-3EL, S/N A01030020 and pressure differential switch P/N 94E42-2 IAW JanAero Devices Service Letters A-101 & A103 and Beech Baron Shop Manual.

2. Removed two sun visors P/N 106-530032-1 and installed Rosen sun visors P/N RBB300-19, S/N 1098084 IAW STC SA3599NM.

-----END-----

Additional Sheets Are Attached

*Customer Copy*

 U.S. Department of Transportation Federal Aviation Administration	<b>MAJOR REPAIR AND ALTERATION</b> <b>(Airframe, Powerplant, Propeller, or Appliance)</b>	Form Approved OMB No. 2120-0020
		For FAA Use Only
		Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act 1958)

1. Aircraft	Make <b>BEECHCRAFT</b>  Serial No. <b>TC-1730</b>	Model <b>B55</b>  Nationality and Registration Mark <b>N42HF</b>
2. Owner	Name (As shown on registration certificate) <b>HUMES, THOMAS M.</b>	Address (As shown on registration certificate) <b>1785 DOOLITTLE CT.          PORT ORANGE, FL. 32128-6757</b>

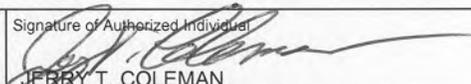
The data identified herein complies with the applicable  
 airworthiness requirements and is approved only for the  
 above described aircraft subject to conformity inspection  
 by a person authorized in FAR 43.7.

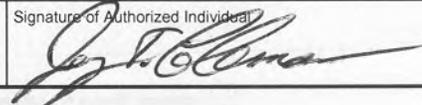
Date APR 20 2004      3. For FAA Use Only      Signature: *Susan M. DeLeon*  
 \_\_\_\_\_ FAA Inspector

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~(As described in item 1 above)~~~~~			<input type="checkbox"/>	<input checked="" type="checkbox"/>
POWERPLANT				<input type="checkbox"/>	<input type="checkbox"/>
PROPELLER				<input type="checkbox"/>	<input type="checkbox"/>
APPLIANCE	Type			<input type="checkbox"/>	<input type="checkbox"/>
	Manufacturer			<input type="checkbox"/>	<input type="checkbox"/>

6. Conformity Statement		
A. Agency's Name and Address <b>DAYTONA AIRCRAFT SERVICES          561 PEARL HARBOR DRIVE          DAYTONA BEACH, FL. 32114</b>	B. Kind of Agency <input type="checkbox"/> U.S. Certified Mechanic <input type="checkbox"/> Foreign Certified Mechanic <input checked="" type="checkbox"/> Certified Repair Station <input type="checkbox"/> Manufacturer	C. Certificate No. <b>DYTR262K</b>

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date <b>April 7, 2004</b>	Signature of Authorized Individual  <b>JERRY T. COLEMAN</b>
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7. Approval for Return to Service			
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED			
BY	<input type="checkbox"/> FAA Fit Standards Inspector <input type="checkbox"/> FAA Designee	<input type="checkbox"/> Manufacturer <input checked="" type="checkbox"/> Repair Station	<input type="checkbox"/> Inspection Authorization <input type="checkbox"/> Person Approved by Transport Canada Airworthiness Group Other (Specify)
Date of Approval or Rejection <b>4-30-04</b>	Certificate or Designation No. <b>DYTR262K</b>	Signature of Authorized Individual 	

## NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

### 8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

REMOVED COMMUNICATIONS CORP. CIR-10 S/N20922 E.L.T.  
INSTALLED IN ITS PLACE AN AMERI-KING AK-450 S/N471062  
EMERGENCY LOCATOR TRANSMITTER PER INSTALL MANUAL #IM-450.  
USING EXISTING ANTENNA.  
IN CONFORMANCE WITH TSO-C91a AND DO-160c

ALL WORK WAS IN ACCORDANCE WITH AC43.13-1B CHAPTER 10,  
CHAPTER 11 SECT. 3-15, CHAPTER 12 SECT. 1,2, AC43.13-2A  
CHAPTER 1 PARA.6,7,9, CHAPTER 2 PARA. 21,22,23,24,27,  
CHAPTER 3 PARA. 36.

IN COMPLIANCE WITH F.A.R. 23.1301 AND 23.1351.

#### INSTRUCTIONS FOR CONTINUED AIRWORTHINESS

AIRCRAFT REG. #: N42HF, EQUIPMENT EMERGENCY LOCATOR TRANSMITTER. FUNCTION: BEACON LOCATOR INTERFACES: N/A  
.OPERATION: REFERENCE APPROPRIATE PILOTS GUIDE SERVICE INFO: ALL SERVICING MUST BE ACCOMPLISHED BY APPROVED  
DEALERS FOR AMERI-KING EQUIPMENT. PROBABLE MALFUNCTIONS: N/A REMOVAL/REPLACEMENT: STANDARD TOOLS FOR  
ACCESS DIAGRAMS: N/A SPECIAL INSPECTIONS: N/A, PROTECTIVE TREATMENTS: N/A, STRUCTURAL DATA: N/A, SPECIAL  
TOOLS: N/A, COMMUTER CATEGORY INFO: N/A, OVERHAUL PERIODS: PERFORM INSPECTION A MINIMUM OF ONE TIME EACH  
YEAR PER IM-450 OPS MANUAL, BATTERY REPLACEMENT IS ON DUE DATE OF BATTERY OR ONE YEAR INTERVALS AND BATTERY  
REPLACEMENT IS REQUIRED WHEN IN USE FOR MORE THAN ONE CUMULATIVE HOUR. AIRWORTHINESS LIMITATIONS: N/A,  
PROCEDURE TO REVISE ICA: A COPY OF A REVISED FORM 337 AND REVISED ICA WILL BE SUBMITTED TO THE LOCAL FSDO  
ALONG WITH A LETTER OF INTENT. ONCE THE FAA APPROVED FORM 337 IS RECEIVED, A MAINTENANCE RECORD WILL BE  
MADE IDENTIFYING THE REVISION, LOCATION, AND DATE OF FORM 337.

WEIGHT AND BALANCE CHANGE NEGLIGIBLE  
AIRCRAFT LOGBOOK ENTRY MADE  
FUNCTIONAL TEST PERFORMED

-----END-----

Additional Sheets Are Attached



US Department of Transportation  
Federal Aviation Administration

## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved  
OMB No. 2120-0020  
2/28/2011

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

<b>1. Aircraft</b>	Nationality and Registration Mark <b>N42HF</b>	Serial No. <b>TC-1730</b>	
	Make <b>Beechcraft</b>	Model <b>95-B55</b>	Series
<b>2. Owner</b>	Name (As shown on registration certificate) <b>Humes, Thomas M.</b>		Address (As shown on registration certificate) Address <b>1785 Doolittle Ct</b>
			City <b>Port Orange</b> State <b>FL</b> Zip <b>32128</b> Country <b>Volusia</b>

### 3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
<input type="checkbox"/>	<input checked="" type="checkbox"/>	AIRFRAME	_____	(As described in Item 1 above)	_____
<input type="checkbox"/>	<input type="checkbox"/>	POWERPLANT			
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type		
			Manufacturer		

### 6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency	
Name <b>Thomas M. Humes</b>		<input checked="" type="checkbox"/> U. S. Certificated Mechanic	Manufacturer
Address <b>1785 Doolittle Ct.</b>		<input type="checkbox"/> Foreign Certificated Mechanic	C. Certificate No.
City <b>Port Orange</b> State <b>FL</b>		<input type="checkbox"/> Certificated Repair Station	
Zip <b>32128</b> Country <b>Volusia</b>		<input type="checkbox"/> Certificated Maintenance Organization	<b>2635991</b>

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual  <b>Thomas M. Humes</b> <b>11 JUN 2009</b>
--	--

### 7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  Approved  Rejected

BY	FAA Flt. Standards Inspector	Manufacturer	Maintenance Organization	Persons Approved by Canadian Department of Transport
	FAA Designee	Repair Station	<input checked="" type="checkbox"/> Inspection Authorization	Other (Specify)

Certificate or Designation No. <b>1278275</b>	Signature/Date of Authorized Individual  <b>M.J. GRAINGER 6-11-2009</b>
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**NOTICE**

*Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.*

**8. Description of Work Accomplished**

*(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)*

N42HF

11 Jun 2009

Nationality and Registration Mark

Date

Installed Airwolf Filter Corp. oil filter kit, P/N AFC-K008 on each engine, TCM IO-550E S/N 681060 (left) and 681061 (right), and on the aircraft firewalls. Engine S/Ns may change in the future. Installation was made in the horizontal configuration IAW STC #SA00079NY and AC43.13-1B/2B. This kit uses Champion P/N 48109 or equivalent oil filters. Instructions for continued airworthiness are included in the STC and shall be part of the aircraft inspection and maintenance requirements. Aircraft weight and balance recomputed and entered into the POH and maintenance records.

-----END-----

Item No.	Model	Make	Part	Alt.	Alt. No.	Alt. Type
	(As described in Item 1 above)		WIRING			<input type="checkbox"/>
			POWERPLANT			<input type="checkbox"/>
			PROPELLER			<input type="checkbox"/>
			APPLANCE			<input type="checkbox"/>

**6. Contractor's Statement**

I certify that the repair and/or alteration made to the aircraft described in Item 2 above and described on the reverse of this certificate complies with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Signature of Authorized Individual: *Thomas M. Haines*  
 Date: *11 Jun 2009*

Signature of FAA Inspector: \_\_\_\_\_  
 Date: \_\_\_\_\_

**7. Approval for Return to Service**

Present to the aircraft crew before specified below the unit identified in Item 2 was inspected in the manner provided by the Administrator of the Federal Aviation Administration and is:

Approved  Rejected

Signature of FAA Inspector: \_\_\_\_\_  
 Date: \_\_\_\_\_

Signature of Authorized Individual: \_\_\_\_\_  
 Date: \_\_\_\_\_

Additional Sheets Are Attached



US Department of Transportation  
Federal Aviation Administration

**MAJOR REPAIR AND ALTERATION**  
**(Airframe, Powerplant, Propeller, or Appliance)**

Form Approved  
OMB No. 2120-0020  
2/28/2011

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark N42HF	Serial No. TC-1730	
	Make Beechcraft	Model 95-B55	Series
2. Owner	Name (As shown on registration certificate) Humes, Thomas M.	Address (As shown on registration certificate) Address 1785 Doolittle Ct	
		City Port Orange	State FL
		Zip 32128	Country Volusia

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
<input type="checkbox"/>	<input checked="" type="checkbox"/>	AIRFRAME	_____	(As described in Item 1 above)	_____
<input type="checkbox"/>	<input type="checkbox"/>	POWERPLANT			
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type		
			Manufacturer		

6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency	
Name Thomas M. Humes	Address 1785 Doolittle Ct. City Port Orange State FL Zip 32128 Country Volusia	<input checked="" type="checkbox"/> U. S. Certificated Mechanic	Manufacturer
		<input type="checkbox"/> Foreign Certificated Mechanic	C. Certificate No.
		<input type="checkbox"/> Certificated Repair Station	2635991
		<input type="checkbox"/> Certificated Maintenance Organization	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual Thomas M. Humes <i>Thomas M. Humes</i> 17 JUN 2010
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7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  Approved  Rejected

BY	FAA Flt. Standards Inspector	Manufacturer	Maintenance Organization	Persons Approved by Canadian Department of Transport
	FAA Designee	Repair Station	<input checked="" type="checkbox"/> Inspection Authorization	Other (Specify)

Certificate or Designation No. 1278275	Signature/Date of Authorized Individual M. J. Grainger <i>M. J. Grainger</i> 6-17-2010
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**NOTICE**

*Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.*

**8. Description of Work Accomplished**

*(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)*

N42HF

17 Jun 2010

Nationality and Registration Mark

Date

Installed Eagle Technologies fuel tank quick drain kit BS-2 IAW STC SA01110CH and Eagle installation instructions 720B, rev M. Instructions for continued airworthiness are included in the STC and shall be part of the aircraft inspection and maintenance requirements. Aircraft weight and balance change negligible.

-----  
END  
-----

Additional Sheets Are Attached



U S Department of  
Transportation  
Federal Aviation  
Administration

## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved  
OMB No. 2120-0020  
**For FAA Use Only**  
Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958)

<b>1. Aircraft</b>	Make <p style="text-align: center;">BEECH</p>	Model <p style="text-align: center;">95-B55(T42A)</p>
	Serial No. <p style="text-align: center;">TC-1730</p>	Nationality and Registration Mark <p style="text-align: center;">N42HF</p>
<b>2. Owner</b>	Name (As shown on registration certificate) <p style="text-align: center;">HUMES, THOMAS M.</p>	Address (As Shown on registration certificate) <p style="text-align: center;">1785 DOOLITTLE COURT DAYTONA BEACH, FLORIDA 32128-6757</p>

### 3. For FAA Use Only

(This section is for FAA use only and contains instructions for repair stations.)

### 4. Unit Identification

Unit	Make	Model	Serial No.	5. Type	
				Repair	Alteration
AIRFRAME	~~~~~ (As described in item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

### 6. Conformity Statement

<b>A. Agency's Name and Address</b>	<b>B. Kind of Agency</b>	<b>C. Certificate No.</b>
QUEST AVIONICS, INC P.O. BOX 770028 OCALA, FL 34477-0028	<input type="checkbox"/> U.S. Certificated Mechanic	QVNR604L
	<input type="checkbox"/> Foreign Certificated Mechanic	
	<input checked="" type="checkbox"/> Certificated Repair Station	
	<input type="checkbox"/> Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date <p style="font-size: 1.2em; font-family: cursive;">February 23, 2011</p>	Signature of Authorized Individual 
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### 7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  APPROVED  REJECTED

<b>BY</b>	FAA Fit. Standards inspector	Manufacturer	Inspection Authorization	Other (Specify)
	FAA Designee	<input checked="" type="checkbox"/> Repair Station	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection <p style="font-size: 1.2em; font-family: cursive;">2/23/2011</p>		Certificate or Designation No. <p style="text-align: center;">QVNR604L</p>	Signature of Authorized Individual 	

## NOTICE

Weight and Balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

### 8. Discription of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

#### REMOVED

- 1 EA UNITED 7040-B4 VERTICAL SPEED INDICATOR.
- 1 EA KING KI-214 VOR/ILS INDICATOR.

#### INSTALLED

- 1 EA ASPEN EFD-1000 PRO ELECTRONIC FLIGHT DISPLAY SYSTEM.

SYSTEM INSTALLED IN ACCORDANCE WITH MANUFACTURER'S INSTALLTION MANUAL P/N900-00003-001, REV.L, AC43.13-1B, CHAPTER 5, sec.1, CHAPTER 11, sec. 4,5,7,10,12,&16,AND CHAPTER 12, AND AC43.13-2B, CHAPTER 2.

SYSTEM INSTALLED in accordance with STC # SA10822SC, REISSUED dated 9/29/2009 rev.2. AIRCRAFT LISTED IN AML DOCUMENT # 035-00007-001 rev.C ON PAGE 10 OF 17.

PDF UNIT MOUNTED IN PILOT'S PANEL AT STATION 65.0 AS PER PARAGRAPH 203 OF AC43.13-2B. RSM MOUNTED IN AFT SECTION ON TOP OF AIRCRAFT AT STATION 182.0. ACU MOUNTED ON BAGGAGE RADIO SHELF AT STATION 45.0 AS PER AC43.13-2B.

SYSTEM CONNECTED TO GARMIN GNS-530W FOR GPS/NAV INFORMATION AND BENDIX/KING KX-165 NAV/COM FOR #2 NAV INFORMATION TO BE DISPLAYED AND INTERFACED WITH BENDIX FCS-810 AUTOPILOT SYSTEM.

EQUIPMENT LIST REVISED TO REFLECT THESE CHANGES.  
AIRCRAFT TO BE WEIGHED.

INSTALLATION COMPLIES WITH FAR 23.1301 AND 23.1431.

TOTAL ELECTRICAL LOAD DOES NOT EXCEED 80% OF TOTAL AIRCRAFT LOAD.

#### INSTRUCTIONS FOR CONTINUED AIRWORTHINESS,

1. Introduction: See above
2. Description of Alteration: See above
3. Control, Operation, information: Refer to ASPEN EFD-1000 Pilot's Guide
4. Servicing Information: N/A
5. Maintenance Instructions: On Condition Only, Periodic maintenance is not required.
6. Troubleshooting Information: Refer to EFD-1000 Maintenance Manual.
7. Removal or Replacement Information: Refer to EFD-1000 Installation Manual.
8. Diagrams: Refer to EFD-1000 Installation Manual.
9. Special Instruction: N/A
10. Applications of Protective Treatments: N/A
11. Data: Relative: N/A
12. Special Tools: N/A
13. Commuter Category: N/A
14. Overhaul Period: N/A
15. Airworthiness Limitation Section: N/A
16. Revision: A letter will be submitted to the local FSDO with a copy of the revised FAA form 337 and revised ICA.

SYSTEM TESTED AND FOUND TO PERFORM SATISFACTORY.

----- END -----

Additional Sheets Are Attached



U S Department of  
Transportation  
Federal Aviation  
Administration

## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved  
OMB No. 2120-0020

**For FAA Use Only**  
Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958)

<b>1. Aircraft</b>	Make <p style="text-align: center;">BEECH</p>	Model <p style="text-align: center;">95-B55 (T42A)</p>
	Serial No. <p style="text-align: center;">TC-1730</p>	Nationality and Registration Mark <p style="text-align: center;">N42HF</p>
<b>2. Owner</b>	Name (As shown on registration certificate) <p style="text-align: center;">HUMES, THOMAS M.</p>	Address (As Shown on registration certificate) <p style="text-align: center;">1785 DOOLITTLE COURT DAYTONA BEACH, FLORIDA 32128-6757</p>

### 3. For FAA Use Only

### 4. Unit Identification

### 5. Type

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in item 1 above) ~~~~~				<b>X</b>
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

### 6. Conformity Statement

A. Agency's Name and Address <p style="text-align: center;">QUEST AVIONICS, INC P.O. BOX 770028 OCALA, FL 34477-0028</p>	B. Kind of Agency <input type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input checked="" type="checkbox"/> Certificated Repair Station <input type="checkbox"/> Manufacturer	C. Certificate No. <p style="text-align: center;">QVNR604L</p>
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D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date <p style="text-align: center;"><i>February 23, 2011</i></p>	Signature of Authorized Individual 
---	--

### 7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  APPROVED  REJECTED

BY	FAA Fit. Standards inspector	Manufacturer	Inspection Authorization	Other (Specify)
	FAA Designee	<input checked="" type="checkbox"/> Repair Station	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection <p style="text-align: center;"><i>2/23/2011</i></p>		Certificate or Designation No. <p style="text-align: center;">QVNR604L</p>	Signature of Authorized Individual 	

## NOTICE

Weight and Balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

### 8. Discription of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

#### REMOVED

- 1 EA KING KX-175B NAV/COM SYSTEM WITH KI-214
- 1 EA NORTHSTAR GPS-600 GPS SYTEM WITH 2006 SERIALIZER.
- 1 EA NAT RS-08-001 RELAY UNIT

#### INSTALLED

- 1 EA GARMIN GNS-530W WAAS GPS/COM/ILS SYSTEM.
- 1 EA GARMIN GA-35 GPS ANTENNA.
- 1 EA DEKLIN DWA-1000 ANNUNCIATOR UNIT.

SYSTEM INSTALLED IN ACCORDANCE WITH AC43.13-1B, CHAPTER 5, sec.1, CHAPTER 11, sec. 2,3,4,5&7, AC43.13-2B, CHAPTERS 2&3, AND AC20-138a, app.1, para.1. INSTALLATION WAS MADE IN REFERENCE WITH MANUFACTURER'S INSTALLATION MANUALS p/n 190-00357-08 rev C dated 6/2010.

SYSTEM INSTALLED I/A/W PREVIOUSLY APPROVED SUPPLEMENTAL TYPE CERTIFICATE (STC) NUMBER SA01933LA dated 11/6/2006. AIRCRAFT LISTED ON APPROVED MANUFACTURER'S LIST ON PAGE 7. PANEL UNIT MOUNTED IN RADIO PANEL AT STATION 61.0 as PER PARAGRAPH 203 OF AC43.13-2B. MOUNTED GPS/WAAS ANTENNA ON TOP AT STATION 118.0 AS PER CHAPTER 3 OF AC43.13-2B. DWA-1000 ANNUNCIATOR MOUNTED IN PILOT'S PANEL AT STATION 67.0. SYSTEM CONNECTED TO EXISTING ASPEN EFD-1000 PRO PFD SYSTEM FOR DISPLAYING NAVIGATION AND GPS INFORMATION. USED EXISTING COM AND NAV ANTENNA SYSTEMS. INSTALLATION COMPLIES WITH FAR 23.1301 AND 23.1431. EQUIPMENT LIST AND WEIGHT AND BALANCE DATA REVISED TO REFLECT THESE CHANGES. ELECTRICAL LOAD ANALYSIS DETERMINED THAT TOTAL ELECTRICAL LOAD DOES NOT EXCEED 80% OF TOTAL AIRCRAFT LOAD.

ELECTROMAGNETIC COMPATIBILITY: THE GNS-530W DOES NOT INTERFERE WITH THE NORMAL OPERATION OF OTHER EQUIPMENT INSTALLED IN THE AIRCRAFT. VERIFIED THAT THERE IS NOT FREQUENCY INTERFERENCE/OR DEGRATION OF NAVIGATIONAL SIGNALS ON THE FOLLOWING VHF FREQUENCIES: 121.150 mhz, 121.175 mhz, 121.200mhz, 131.250mhz, AND 131.300mhz.

#### INSTRUCTIONS FOR CONTINUED AIRWORTHINESS,

1. Introduction: See ABOVE
2. Description of Alteration: See Above
3. Control, Operation, information: Refer to Garmin GNS-530W Pilot's Guide
4. Servicing Information: N/A
5. Maintenance Instructions: On Condition Only, Periodic maintenance is not required.
6. Troubleshooting Information: Refer to GNS-530W Maintenance Manual.
7. Removal or Replacement Information: Refer to GNS-530W Installation Manual.
8. Diagrams: Refer to GNS-530W Installation Manual.
9. Special Instruction: N/A
10. Applications of Protective Treatments: N/A
11. Data: Relative: N/A
12. Special Tools: N/A
13. Commuter Category: N/A
14. Overhaul Period: N/A
15. Airworthiness Limitation Section: N/A
16. Revision: A letter will be submitted to the local FSDO with a copy of the revised FAA form 337 and revised ICA.

SYSTEM FUNCTION CHECKED AND FOUND TO PERFORM SATISFACTORY. THE GARMIN GNS-530W GPS APPROVED FOR VFR AND IFR ENROUTE, TERMINAL, AND PRECISION APPROACH NAVIGATION AND WAS INSTALLED AND TESTED IN ACCORDANCE WITH AC20-138A, PARAGRAPH 8.c.(2).

FAA APPROVED FLIGHT MANUAL SUPPLEMENTAL (FMS) INSTALLED IN AIRCRAFT'S APPROVED FLIGHT MANUAL (AFM).

AIRCRAFT PLACARDED "GPS APPROVED FOR VFR AND IFR ENROUTE, TERMINAL, AND PRECISION APPROACHES".

----- END -----

Additional Sheets Are Attached



U S Department of  
Transportation  
Federal Aviation  
Administration

## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved  
OMB No. 2120-0020  
**For FAA Use Only**  
Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958)

<b>1. Aircraft</b>	Make <p style="text-align: center;">BEECH</p>	Model <p style="text-align: center;">95-B55(T42A)</p>
	Serial No. <p style="text-align: center;">TC-1730</p>	Nationality and Registration Mark <p style="text-align: center;">N42HF</p>
<b>2. Owner</b>	Name (As shown on registration certificate) <p style="text-align: center;">HUMES, THOMAS M</p>	Address (As Shown on registration certificate) <p style="text-align: center;">1785 DOOLITTLE COURT PORT ORANGE, FLORIDA 32128-6757</p>

### 3. For FAA Use Only

(This section is for FAA use only and contains instructions for repair stations.)

### 4. Unit Identification

### 5. Type

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	----- (As described in item 1 above) -----				<b>X</b>
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

### 6. Conformity Statement

<b>A. Agency's Name and Address</b>  QUEST AVIONICS, INC P.O. BOX 770028 OCALA, FL 34477-0028	<b>B. Kind of Agency</b> <input type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input checked="" type="checkbox"/> Certificated Repair Station <input type="checkbox"/> Manufacturer	<b>C. Certificate No.</b>  <p style="text-align: center;">QVNR604L</p>
---	---	--

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date <p style="text-align: center;"><i>February 13, 2015</i></p>	Signature of Authorized Individual 
---	--

### 7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  APPROVED  REJECTED

<b>BY</b>	FAA Fit. Standards inspector	Manufacturer	Inspection Authorization	Other (Specify)
	FAA Designee	<input checked="" type="checkbox"/> Repair Station	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection <p style="text-align: center;"><i>2/13/2015</i></p>		Certificate or Designation No. <p style="text-align: center;">QVNR604L</p>	Signature of Authorized Individual 	

## NOTICE

Weight and Balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

### 8. Discription of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

#### REMOVED

1 EA KING KT-76 TRANSPONDER.

#### INSTALLED

1 EA BENDIX/KING KT-74 MODE S TRANSPONDER SYSTEM WITH ADS-B OUT.

SYSTEM INSTALLED IN ACCORDANCE WITH AC43.13-1B, CHAPTER 5, sec.1, CHAPTER 11, sec. 2,3,4,5&7, AND AC43.13-2B, CHAPTERS 2&3.

INSTALLATION WAS MADE IN REFERENCE WITH MANUFACTURER'S INSTALLATION MANUALS p/n D201308000037, DATED 11/15/2013 REV. D.

SYSTEM INSTALLED I/A/W PREVIOUSLY APPROVED SUPPLEMENTAL TYPE CERTIFICATE (STC) NUMBER SA00765DE. AIRCRAFT LISTED ON APPROVED MANUFACTURER'S LIST ON PAGE 9 OF 30.

PANEL UNIT MOUNTED IN RADIO STACK AT STATION 65.0, WHERE REMOVED UNIT WAS, AS PER PARAGRAPH 203 OF AC43.13-2B. USED EXISTING ANTENNA SYSTEM.

SYSTEM CONNECTED TO EXISTING GNS-530W DISPLAY AND INFORMATION FOR ADS-B OUT AND EXISTING AUDIO SYSTEM.

INSTALLATION COMPLIES WITH FAR 23.1301 AND 23.1431.

EQUIPMENT LIST AND WEIGHT AND BALANCE DATA REVISED TO REFLECT THESE CHANGES.

ELECTRICAL LOAD ANALYSIS DETERMINED THAT TOTAL ELECTRICAL LOAD DOES NOT EXCEED 80% OF TOTAL AIRCRAFT LOAD.

### INSTRUCTIONS FOR CONTINUED AIRWORTHINESS,

1. Introduction: See ABOVE
2. Description of Alteration: See Above
3. Control, Operation, information: Refer to BENDIX/KING KT-74 Pilot's Guide
4. Servicing Information: N/A
5. Maintenance Instructions: On Condition Only, Periodic maintenance is not required.
6. Troubleshooting Information: Refer to KT-74 Maintenance Manual.
7. Removal or Replacement Information: Refer to KT-74 Installation Manual.
8. Diagrams: Refer to KT-74 Installation Manual.
9. Special Instruction: N/A
10. Applications of Protective Treatments: N/A
11. Data: Relative: N/A
12. Special Tools: N/A
13. Commuter Category: N/A
14. Overhaul Period: N/A
15. Airworthiness Limitation Section: N/A
16. Revision: A letter will be submitted to the local FSDO with a copy of the revised FAA form 337 and revised ICA.

SYSTEM FUNCTION CHECKED AND FOUND TO PERFORM SATISFACTORY.

FAA APPROVED FLIGHT MANUAL SUPPLEMENTAL (FMS) INSTALLED IN AIRCRAFT'S APPROVED FLIGHT MANUAL (AFM).

----- END -----

Additional Sheets Are Attached

SEND TO FAA OR City



US Department of Transportation  
Federal Aviation Administration

**MAJOR REPAIR AND ALTERATION  
(Airframe, Powerplant, Propeller, or Appliance)**

Form Approved  
OMB No. 2120-0020  
2/28/2011

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark <b>N42HF</b>	Serial No. <b>TC-1730</b>
	Make <b>BEECH</b>	Model <b>BARON</b> Series <b>95-B55</b>
2. Owner	Name (As shown on registration certificate) <b>THOMAS HUMES</b>	Address (As shown on registration certificate) Address <b>1785 DOOLITTLE CT</b> City <b>PORT ORANGE</b> State <b>FL</b> Zip <b>32128</b> Country <b>USA</b>

**3. For FAA Use Only**

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
<input type="checkbox"/>	<input type="checkbox"/>	AIRFRAME	_____	(As described in Item 1 above)	_____
<input type="checkbox"/>	<input type="checkbox"/>	POWERPLANT			
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type		
			Manufacturer		

**6. Conformity Statement**

A. Agency's Name and Address		B. Kind of Agency	
Name <b>STRUCTURAL REPAIR SPECIALISTS</b>	Address <b>5901 217 ST</b>	City <b>PRIOR LAKE</b> State <b>MN</b>	Zip <b>55372</b> Country <b>US</b>
		<input type="checkbox"/> U. S. Certificated Mechanic	<input type="checkbox"/> Manufacturer
		<input type="checkbox"/> Foreign Certificated Mechanic	<input type="checkbox"/> C. Certificate No.
		<input checked="" type="checkbox"/> Certificated Repair Station	
		<input type="checkbox"/> Certificated Maintenance Organization	<b>U5LRO68X</b>

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual  <b>19 MAR 15</b>
--	---

**7. Approval for Return to Service**

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  Approved  Rejected

BY	FAA Fit. Standards Inspector	Manufacturer	Maintenance Organization	Persons Approved by Canadian Department of Transport
	FAA Designee <input checked="" type="checkbox"/>	Repair Station	Inspection Authorization	Other (Specify)

Certificate or Designation No. <b>U5LRO68X</b>	Signature/Date of Authorized Individual  <b>19 MAR 15</b>
---	---

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

**8. Description of Work Accomplished**

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N42HF

19 MAR 15

Nationality and Registration Mark

Date

OVERHAULED RH ELEVATOR EXCEPT TRIM TAB BY TREATING AND PRIMING FRAME AND REPLACING ALL MAGNESIUM PARTS USING SRS AVIATION LLC ALUMINUM SKIN KIT I/A/W STC SA02172CH. ALUMINUM FITTINGS INSTALLED COMPLY WITH AD 89-05-02. ELEVATOR PAINTED AND BALANCED \_\_\_\_\_ RH. ELEVATOR NOT INSTALLED BY SRS.

UPDATED AIRCRAFT WEIGHT AND BALANCE PER ICA PAGE 9

CERT#: 2635991 NAME: Thomas Humes DATE 29 Mar 2015

Additional Sheets Are Attached



US Department of Transportation  
Federal Aviation Administration

**MAJOR REPAIR AND ALTERATION**  
**(Airframe, Powerplant, Propeller, or Appliance)**

OMB No. 2120-0020  
Exp: 5/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark N42HF	Serial No. TC-1730
	Make Beechcraft	Model 95-B55
2. Owner	Name (As shown on registration certificate) Humes, Thomas M.	
	Address (As shown on registration certificate) Address 1785 Doolittle Ct	
	City Port Orange	State FL
	Zip 32128	Country USA

**3. For FAA Use Only**

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
<input type="checkbox"/>	<input checked="" type="checkbox"/>	AIRFRAME	Beechcraft	(As described in Item 1 above)	TC-1730
<input type="checkbox"/>	<input type="checkbox"/>	POWERPLANT			
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type		
			Manufacturer		

**6. Conformity Statement**

A. Agency's Name and Address		B. Kind of Agency	
Name Thomas Humes	Address 1785 Doolittle Ct City Port Orange State FL Zip 322128 Country USA	<input checked="" type="checkbox"/> U. S. Certificated Mechanic	Manufacturer
		<input type="checkbox"/> Foreign Certificated Mechanic	C. Certificate No.
		<input type="checkbox"/> Certificated Repair Station	2635991 <i>KLP</i>
		<input type="checkbox"/> Certificated Maintenance Organization	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual <i>[Signature]</i> 10 Aug 18
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**7. Approval for Return to Service**

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  Approved  Rejected

BY	FAA Fit. Standards Inspector	Manufacturer	Maintenance Organization	Persons Approved by Canadian Department of Transport
	FAA Designee	Repair Station	<input checked="" type="checkbox"/> Inspection Authorization	Other (Specify)

Certificate or Designation No. 1525270IA	Signature/Date of Authorized Individual <i>[Signature]</i> 10 AUG 2018
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**NOTICE**

*Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.*

**8. Description of Work Accomplished**

*(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)*

N42HF

10 Aug 2018

Nationality and Registration Mark

Date

Installed Marsh Brothers Baron NLG door hinge bushing kit M1005B IAW STC SA03853NY and Installation Instruction II0005. Maintenance Manual Supplement MMS0005 shall be included in the aircraft inspection and maintenance requirements.

-----End-----

Category	Item	Make	Unit	Approved	Checked
			APLANCE	<input type="checkbox"/>	<input type="checkbox"/>
			PHOENIX	<input type="checkbox"/>	<input type="checkbox"/>
			POWER	<input type="checkbox"/>	<input type="checkbox"/>
			ALTR RAME	<input checked="" type="checkbox"/>	<input type="checkbox"/>

4. Company Statement  
 I certify that the work and/or alteration made in the aircraft described in this report was done in accordance with the applicable Federal Aviation Regulations and the information furnished herein is true and correct to the best of my knowledge.

5. Approval for Return to Service  
 Pursuant to the Airframe, Propeller, and Powerplant sections below, the aircraft described in this report was inspected in the aircraft plant and by the Administrator of the Federal Aviation Administration, and it is approved for return to service.

BY:  Repair Station     FAA Repair Station Inspector

Additional Sheets Are Attached



US Department  
of Transportation  
Federal Aviation  
Administration

**MAJOR REPAIR AND ALTERATION**  
**(Airframe, Powerplant, Propeller, or Appliance)**

OMB No. 2120-0020  
Exp: 5/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark N42HF	Serial No. TC-1730
	Make Beechcraft	Model 95-B55
2. Owner	Name (As shown on registration certificate) Humes, Thomas M.	
	Address (As shown on registration certificate) Address 1785 Doolittle Ct	
	City Port Orange	State FL
	Zip 32128	Country USA

**3. For FAA Use Only**

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
<input type="checkbox"/>	<input checked="" type="checkbox"/>	AIRFRAME	Beechcraft	(As described in Item 1 above)	TC-1730
<input type="checkbox"/>	<input type="checkbox"/>	POWERPLANT			
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type Manufacturer		

**6. Conformity Statement**

A. Agency's Name and Address		B. Kind of Agency		C. Certificate No.	
Name Thomas Humes	Address 1785 Doolittle Ct City Port Orange State FL Zip 322128 Country USA	<input checked="" type="checkbox"/>	U. S. Certificated Mechanic	2635991	
		<input type="checkbox"/>	Foreign Certificated Mechanic		
		<input type="checkbox"/>	Certificated Repair Station		
		<input type="checkbox"/>	Certificated Maintenance Organization		

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual <i>[Signature]</i> 30 Apr 19
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**7. Approval for Return to Service**

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  Approved  Rejected

BY	FAA Fit. Standards Inspector	Manufacturer	Maintenance Organization	Persons Approved by Canadian Department of Transport
	FAA Designee	Repair Station	<input checked="" type="checkbox"/> Inspection Authorization	Other (Specify)

Certificate or Designation No. 1525270IA	Signature/Date of Authorized Individual <i>Conrad P. Shub</i> 30 April 2019
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NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N42HF

10 Aug 2018

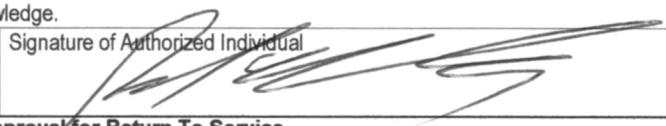
Nationality and Registration Mark

Date

Installed Marsh Brothers Baron MLG door hinge bushing kit M1001, S/N 77-05 IAW STC SA03342NY and Installation Instruction II0002. Maintenance Manual Supplement MMS0002 shall be included in the aircraft inspection and maintenance requirements.

-----End-----

Additional Sheets Are Attached

 U S Department of Transportation Federal Aviation Administration		<b>MAJOR REPAIR AND ALTERATION</b> <b>(Airframe, Powerplant, Propeller, or Appliance)</b>		Form Approved OMB No. 2120-0020 <b>For FAA Use Only</b> Office Identification		
INSTRUCTIONS: Print or type all entries. See FAR 43.9 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958)						
<b>1. Aircraft</b>		Make BEECH	Model 95-B55 (42A)			
		Serial No. TC-1730	Nationality and Registration Mark N42HF USA			
<b>2. Owner</b>		Name (As shown on registration certificate) HUMES, THOMAS M		Address (As Shown on registration certificate) 1785 DOOLITTLE COURT PORT ORANGE, FLORIDA 32128-6757		
<b>3. For FAA Use Only</b>						
<b>4. Unit Identification</b>						
					<b>5. Type</b>	
Unit	Make	Model	Serial No.	Repair	Alteration	
AIRFRAME	(As described in item 1 above)				<b>X</b>	
POWERPLANT						
PROPELLER						
APPLIANCE	Type					
	Manufacturer					
<b>6. Conformity Statement</b>						
A. Agency's Name and Address		B. Kind of Agency		C. Certificate No.		
QUEST AVIONICS, INC P.O. BOX 770028 OCALA, FL 34477-0028		<input type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input checked="" type="checkbox"/> Certificated Repair Station <input type="checkbox"/> Manufacturer		QVNR604L		
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.						
Date		Signature of Authorized Individual				
<i>March 10, 2021</i>						
<b>7. Approval for Return To Service</b>						
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED						
BY	FAA Fit. Standards inspector	Manufacturer	Inspection Authorization	Other (Specify)		
	FAA Designee	<input checked="" type="checkbox"/> Repair Station	Person Approved by Transport Canada Airworthiness Group			
Date of Approval or Rejection		Certificate or Designation No.	Signature of Authorized Individual			
<i>3/10/2021</i>		QVNR604L				

## NOTICE

Weight and Balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

### 8. Discription of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

REMOVED  
1 EA BENDIX/KING KCS-55A SYSTEM

INSTALLED  
1 EA GARMIN GI-275 MULTI-FUNCTION DISPLAY UNIT.

SYSTEM INSTALLED IN ACCORDANCE WITH AC43.13-1B, AC43.13-2B, CHAPTER 2, AND IN REFERENCE WITH MANUFACTURER'S INSTALLATION MANUAL P/N 190-002246-03 & 190-02246-10 (LATEST REVISIONS) AND SA02658SE DATED 01/13/2020. INSTALLATION OF THIS SYSTEM INCLUDED IN FAA APPROVED MODEL LIST (AML) ON PAGE 30 OF 36.

GI-275 UNIT MOUNTED IN PILOT'S SUB PANEL AT STATION 64.0 AS PER PARAGRAPH 203, CHAPTER 2 OF AC43.13-2B. CONNECTED TO EXISTING GARMIN GNS-530W AND BENDIX/KING KX-165. INSTALLATION COMPLIES WITH FAR 23.2505 AND 23.2510.

EQUIPMENT LIST AND WEIGHT AND BALANCE DATA WAS REVISED TO REFLECT THESE CHANGES.

ELECTRICAL LOAD ANALYSIS DETERMINED THAT TOTAL ELECTRICAL LOAD DOES NOT EXCEED 80% OF TOTAL AIRCRAFT LOAD.

ELECTROMAGNETIC COMPATIBILITY: THE GI-275 DOES NOT INTERFERE WITH THE NORMAL OPERATION OF OTHER EQUIPMENT INSTALLED IN THE AIRCRAFT. VERIFIED THAT THERE IS NOT FREQUENCY INTERFERENCE/OR DEGRATION OF NAVIGATIONAL SIGNALS ON THE FOLLOWING VHF FREQUENCIES; 121.150 Mhz, 121.175 Mhz, 121.200 Mhz, 131.250 Mhz and 131.300 Mhz.

#### INSTRUCTIONS FOR CONTINUED AIRWORTHINESS;

1. Introduction: see ABOVE
2. Description of Alteration: see ABOVE
3. Control, operation, information: Refer to Garmin GI-275 Pilot's Guide
4. Servicing Information: N/A
5. Maintenance Instructions: On Condition only, Periodic maintenance is not required.
6. Troubleshooting Information: Refer to GI-275 Maintenance Manual.
7. Removal or Replacement Information: Refer to GI-275 Installation Manual.
8. Diagrams: Refer to GI-275 Installation manual.
9. Special Instructions: N/A
10. Applications of Protective Treatments: N/A
11. Data: Relative: N/A
12. Special Tools: N/A
13. Commuter category: N/S
14. Overhaul Period: N/A
15. Airworthiness Limitation section: N/A
16. Revision: A letter will be submitted to the Aircraft Registration Branch (AFS-750) in Oklahoma City. AN entry in the aircraft records should indicate the current revision.

FLIGHT MANUAL SUPPLEMENT (FMS) INSTALLED IN AIRCRAFT OPERATION MANUAL.

SYSTEM FUNCTION CHECKED AND FOUND TO PERFORM SATISFACTORY.

-----END-----

Additional Sheets Are Attached